



# Development and Control of Novel Vibration Isolation Platform

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**Abstract:** A vibration isolation platform designed to attenuate high frequency vibration is proposed consisting of an active layer driven by a piezoelectric actuator and a passive layer. The proposed platform can achieve vibration isolation both in the vertical and horizontal directions. The dynamic behavior of the system is analyzed by deriving the equations of motions using the Lagrangian approach. System identifications are performed to validate the plant dynamics. The robust  $H_\infty$  controller is adopted to design the controller with consideration of payload uncertainties. Experimental results show that the control can reduce vibrations by an average of 10dB within a frequency range of 15-40Hz.

**Keywords:** Vibration isolation; pendulum structure; active-passive;  $H_\infty$  controller

## Introduction

The importance of vibration isolation is increasing as manufacturing technology moves toward the nanoscale. Passive, active and active-passive techniques have been developed to protect high precision equipment from ground disturbance. Passive vibration isolation systems use an oil channel, pneumatic device or damping layer to reduce the transmission of seismic vibration. The passive approach uses a low first resonance frequency with a high mass and low stiffness, such as an optical table. The frequency response between the magnitude of the payload and ground disturbance could have a 20dB/decade attenuation after passing the first natural frequency. However the passive isolators could result in the vibrations around the resonance being amplified rather than isolated. Using a high damping ratio can reduce the vibration around the resonance, but this could reduce the attenuation rate after the resonance. Applying active controls can provide more vibration attenuation around the resonance. Aside from the critical passive design problem, passive systems

can only handle light payloads due to the low stiffness design [1]. The semi-active system was proposed to control stiffness and damping, using a damper design proposed by [2] or the Magnetorheological damper [3].

During the past two decades, active and active-passive systems using piezoelectric actuators have been widely studied [4, 5]. The active-passive system generally consists of an active layer and a passive layer to achieve advantages such as stability, energy savings, and robustness. Many control techniques have been proposed including classical control [6, 7]. Optimal LQG control [8] adaptive control [9], and neural networks [10]. Recently, six degree of freedom systems such as the Stewart platform [11] have been proposed. The cross coupling effects between each leg were investigated [8, 12] using either the singular value decomposition method or modal decompositions.

Although many designs have been proposed to achieve active-passive vibration isolation systems, such designs were only focused on reducing the transmissibility in the vertical direction without considering the horizontal direction. This paper proposes a novel pendulum-typed system consisting of three



active cables embedded with piezoelectric actuators. The proposed design can passively attenuate horizontal vibrations while its active actions attenuate vertical vibrations. Theoretical modeling and experiments are conducted to validate the design concept. A robust controller based on  $H_\infty$  theory is then proposed to deal with payload uncertainties. Experiments are conducted to validate the controller performance.

### Equations of Vertical and Horizontal Motion

Figure 1 illustrates the interior design of the pendulum-type vibration isolation system. The system comprises three cables connecting the outer case and intermediate (center) mass. The piezoelectric actuator and accelerometer are embedded in the intermediate mass to form the active layer. On top of the intermediate mass, an isolation pad and payload disk form the passive layer. To test the isolation performance, the system is set on a marble table. The proposed system can provide active control for low-frequency vibrations and a passive attenuation of high-frequency vibrations due to the low first natural frequency of the pendulum. The closed loop control picks up the vibration signal from the accelerometer and then drives the piezoelectric actuators based on the designed control algorithm. The low frequency vibration transmissibility is first reduced by the first active layer. The high frequency vibrations can be then further reduced by the passive layer. Because the vibration around the resonance cannot be reduced passively, the active action can provide the isolation around this region to achieve the advantages of both the active and passive effects.

#### Dynamic Equation in the Vertical Direction

To derive the dynamic model, the schematic of the vibration system can be represented by the lumped parameter system in the vertical direction as shown in Figure 2.

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Generally, the force of the piezoelectric actuator can be represented as [13]

$$F_a = -J_c V k_a \Delta X, \tag{1}$$

where  $k_a$  is the stiffness of the piezoelectric actuators,  $J_c$  is the driving coefficient, and  $\Delta X$  is the displacement of the actuators.

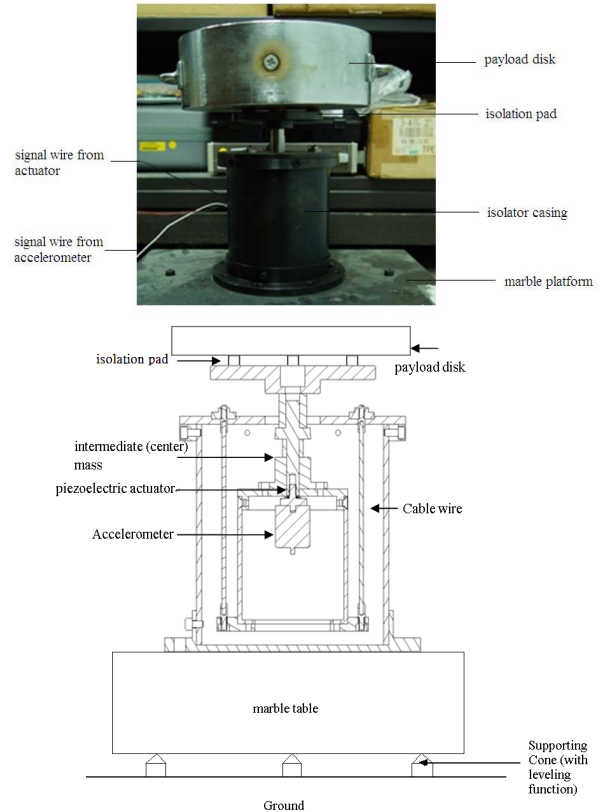


Figure 1. Schematic illustrations of vibration isolation platform.

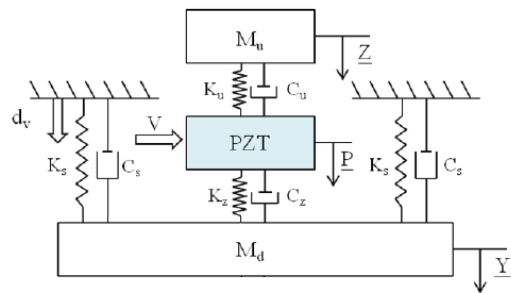


Figure 2. Lumped parameter system representing the vibration isolation system in the vertical direction.

The dynamic equation can be derived by using the energy method with the Lagrangian approach. First, the kinetic energy  $T_a$ , potential energy  $V_a$  and virtual work in the vertical direction shown in Figure 2 can be expressed as:

$$T_a = \frac{1}{2} M_u \dot{Z}^2 + \frac{1}{2} M_d \dot{Y}^2, \tag{2}$$

$$V_a = \frac{1}{2}K_u(P-Z)^2 + \frac{1}{2}K_z(Y-P)^2 + \frac{1}{2}K_s(Y-dv)^2, \quad (3)$$

$$\begin{aligned} \delta W &= F\delta z - C_u(\dot{P} - \dot{Z})\delta(P-Z) - \\ &C_z(\dot{Y} - \dot{P})\delta(Y-P) - C_s(\dot{Y} - \dot{dv})\delta(Y-dv). \end{aligned} \quad (4)$$

By applying the Lagrange formulation given in Eq. (5), the dynamic equations can be formulated as Eq. (6).

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_i} - \frac{\partial L}{\partial q} = Q_i, \quad (5)$$

$$\begin{aligned} &\begin{bmatrix} M_u & 0 & 0 \\ 0 & M_d & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \ddot{Z} \\ \ddot{Y} \\ \ddot{P} \end{bmatrix} + \begin{bmatrix} C_u & 0 & -C_u \\ 0 & C_z + C_s & -C_z \\ -C_u & -C_z & C_u + C_z \end{bmatrix} \begin{bmatrix} \dot{Z} \\ \dot{Y} \\ \dot{P} \end{bmatrix} \\ &+ \begin{bmatrix} K_u & 0 & -K_u \\ 0 & K_z + K_s & -K_z \\ -K_u & -K_z & K_u + K_z \end{bmatrix} \begin{bmatrix} Z \\ Y \\ P \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ C_s \\ K_s \end{bmatrix} \dot{d}_v + \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} d_v. \end{aligned} \quad (6)$$

Here,  $M_u$  and  $M_d$  respectively represent the upper and intermediate masses, while  $dv$  is the vertical disturbance displacement.

#### Dynamic Equation in the Horizontal Direction

The horizontal dynamic equation can be represented by the simplified lumped parameter system as shown in Figure 3. Here  $d_h$  is the horizontal disturbance displacement,  $X_T$  is the corresponding displacement of the upper mass in the horizontal direction, and  $\theta$  is the angle of the cables with respect to the vertical direction.

Based on the assumption that the rotational angle is small and the elongation of the cable can be negligible, the kinetic energy  $T_h$ , potential energy  $V_h$  and virtual work in the horizontal direction can be represented as:

$$T_h = \frac{1}{2}M_u\dot{X}_T^2 + \frac{1}{2}M_d\left(L_w \frac{d\sin\theta_w}{dt} + \dot{d}_h\right)^2, \quad (7)$$

$$V_h = \frac{1}{2}K_{uh}(X_T - L_w \sin\theta_w - d_h)^2 + g(M_u + M_d)(L_w - L_w \sin\theta_w), \quad (8)$$

$$\begin{aligned} \delta W_h &= C_{uh}\left(\dot{X}_T - L_w \frac{d\sin\theta_w}{dt} - \dot{d}_h\right)\delta(X_T - L_w \sin\theta_w - d_h) \\ &+ C_{dh}\left(L_w \frac{d\sin\theta_w}{dt}\right)\delta(L_w \sin\theta_w). \end{aligned} \quad (9)$$

By applying the Lagrange equation, the dynamic equations in the horizontal direction are expressed as:

$$\begin{aligned} &\begin{bmatrix} M_u & 0 \\ 0 & (M_u + M_d)L_w^2 \end{bmatrix} \begin{bmatrix} \ddot{X}_T \\ \ddot{\theta}_w \end{bmatrix} + \begin{bmatrix} C_{uh} & -C_{uh}L_w \\ -C_{uh}L_w & (C_{uh} + C_{dh})L_w^2 \end{bmatrix} \begin{bmatrix} \dot{X}_T \\ \dot{\theta}_w \end{bmatrix} \\ &+ \begin{bmatrix} K_{uh} & -K_{uh}L_w \\ -K_{uh}L_w & (M_u + M_d)L_w g + K_{uh}L_w^2 \end{bmatrix} \begin{bmatrix} X_T \\ \theta_w \end{bmatrix} \\ &= \begin{bmatrix} 1 \\ 0 \end{bmatrix} F + \begin{bmatrix} 0 \\ -(M_u + M_d)L_w \end{bmatrix} \ddot{d}_h + \begin{bmatrix} C_{uh} \\ -C_{uh}L_w \end{bmatrix} \dot{d}_h + \begin{bmatrix} K_{uh} \\ -K_{uh}L_w \end{bmatrix} d_h \end{aligned} \quad (10)$$

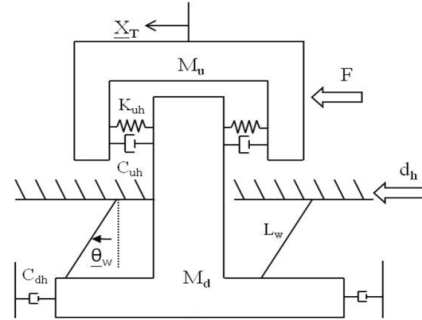


Figure 3. Lumped parameter representation of vibration isolation system in horizontal direction.

### Experimental Validation and Analysis of the Dynamic Model

This section presents the validation and analysis of the dynamic model by comparing the experimental and simulation results. The frequency response of the vertical model can be obtained by applying a chirp sine signal to the piezoelectric actuator and obtaining the output signal from the accelerometers embedded in the intermediate mass using the dynamic signal analyzer HP35670A. Figure 4 shows the simulation and experimental results are in good agreement. Next, the model in the horizontal direction can be validated by hitting the upper mass with a hammer and then measuring the signal from the accelerometer mounted on the upper mass. The frequency response function between  $X_T$  and  $F$  in Figure 3 is shown in Figure 5.

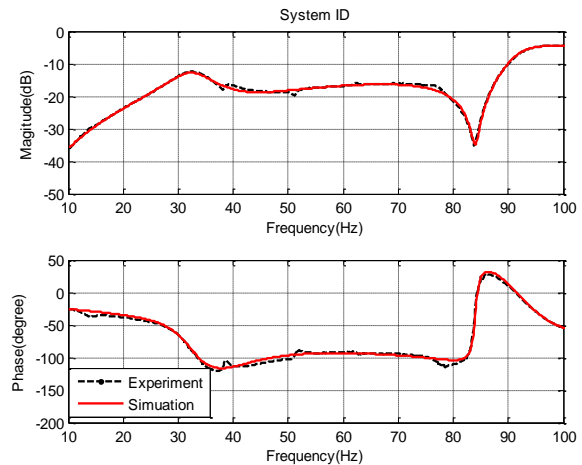


Figure 4. Simulation and experimental results for frequency response function in the vertical direction.

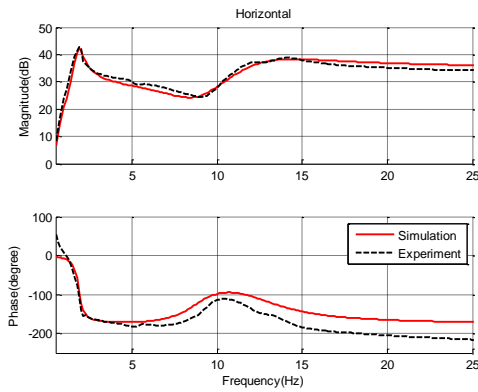


Figure 5. Simulation and experimental results for frequency response function in the horizontal direction.

Because the vibration in the horizontal direction should be attenuated passively, the frequency response function between the disturbance  $d_h$  and  $X_T$  is important for the evaluating the passive transmissibility, as shown in Figure 6. The results demonstrate that the pendulum causes low resonance at about 1.65Hz. After the first resonance the isolation become effective until the second resonance at about 12Hz. The second resonance is mainly caused by the passive isolation pad and the mass effect from the intermediate and upper mass. The horizontal disturbance can obtain an attenuation of at least 5 dB above 16.5Hz. Any disturbance consisting of a frequency above 20Hz can be reduced by more than 10 dB. The only tradeoff is that the input is amplified when the frequency is lower than 2 Hz, mainly due to the low damping of the pendulum system, but this can be solved by adding oil in the case. The oil provides a resistive force to the motion of the intermediate mass, thus increasing the damping force. The second approach is to attach another damper between the outer case and the intermediate mass. Both methods can increase the damping  $C_{dh}$  in Figure 3. However, as mentioned above, the passive control could amplify the resonance even with a high damping design. This drawback can be improved through the use of active controls.

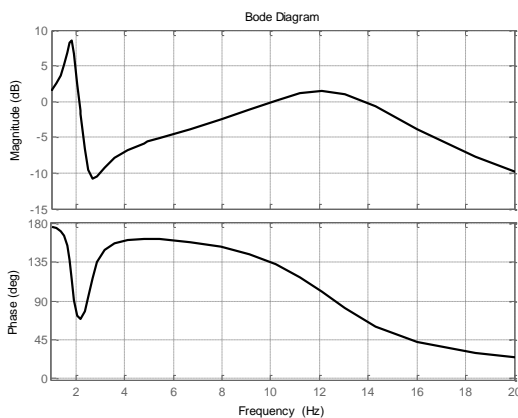


Figure 6 Simulation results for transmissibility in the horizontal direction.

## Robust $H_\infty$ Control of the Platform

After obtaining the frequency response of the plant shown in Figure 4, the next step is to perform the active control for the isolation platform in the vertical direction. In real applications, the weight of the payload set on the vibration platform could vary significantly. To account for the payload uncertainty, this study adopts the robust  $H_\infty$  control.

Figure 7 shows the block diagram used in designing the robust controller where  $d$  is the disturbance,  $P(s)$  is the plant, and  $K(s)$  is the controller.  $W_1(s)$  and  $W_2(s)$  are the performance weighting and uncertainty weighting functions, respectively. Here the multiplicative uncertainty model is used to describe the plant uncertainty under different payloads. As shown in Figure 7, the multiplicative uncertainty can be described as:

$$\Delta P(s) = \frac{P(s) - P_0(s)}{P_0(s)} \quad (11)$$

The nominal plant  $P_0(s)$  is defined as a payload equal to 50 Kg. One of the disturbed plant  $P(s)$  is set as a plant with a payload equal to 40 Kg. The other has a payload equal to 60 Kg. The weighting function  $W_2(s)$  is chosen to cover the uncertainty of  $\Delta P(s)$ . As shown in Figure 8, the  $\Delta P(s)$  of the frequency response functions for the plants with different payloads are measured and  $W_2(s)$  is chosen to cover the uncertainty.

$$W_2(s) = \frac{4s + 450}{s + 400} \quad (12)$$

The performance weighting function  $W_1(s)$  is chosen as a low pass filter which weighs heavily at the target frequencies from 10Hz to 50Hz. The frequency responses of  $W_1(s)$  are shown in Figure 9. After the weighting functions are selected, the block diagram in Figure 9 can be expressed in the form of the linear fractional transformation shown in Figure 10. Here, the augmented plant  $P_a$  is given as:

$$\begin{bmatrix} z_1 \\ z_2 \\ e \end{bmatrix} = \begin{bmatrix} -W_1 & -W_2 & -W_1 P \\ 0 & 0 & W_2 P \\ -1 & -1 & -P \end{bmatrix} \begin{bmatrix} d_1 \\ d_2 \\ u \end{bmatrix}, \quad (13)$$

and

$$\Delta = \begin{bmatrix} \delta_1 & 0 \\ 0 & \delta_m \end{bmatrix}. \quad (14)$$

$\delta_1$  and  $\delta_m$  are  $1 \times 1$  diagonal blocks reflecting the nominal performance and robust stability respectively.  $\Delta$  is normalized as  $\|\Delta\|_\infty < 1$ . With the given augmented plant, the  $H_\infty$  controller  $K(s)$  can be obtained easily by solving two Riccati equations [14, 15].

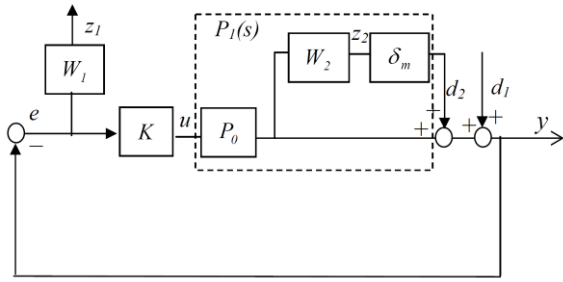


Figure 7. Block diagram of the controlled system with plant uncertainty.

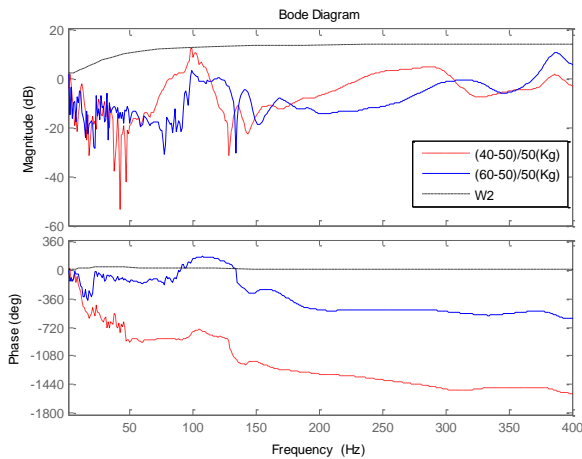


Figure 8. Frequency response of the plant uncertainty.

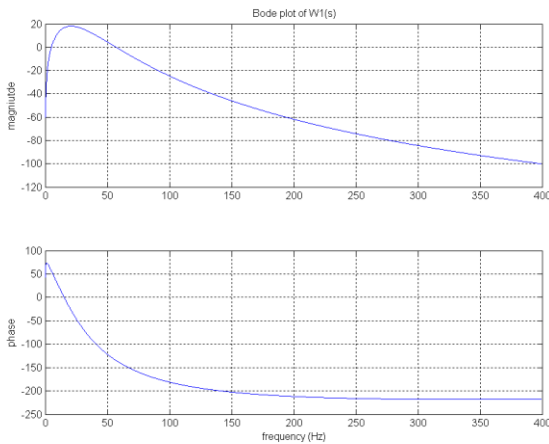


Figure 9. Frequency response functions of the weighting function  $W1(s)$ .

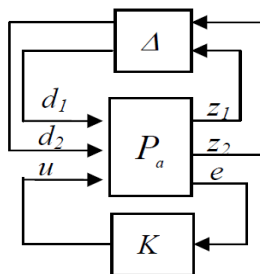


Figure 10. A linear fractional representation of the control system.

After designing the controller, experiments were conducted to validate system performance. From the performance weighting shown in Figure 9, it is found that

higher weightings are set within a frequency range from 10 to 50 Hz. The controller is then implemented on a DSP-based controller with an A/D sampling equal to 1 KHz. The disturbance can be simulated by generating a chirp sine signal injected into the plant at a control voltage  $u$ , with results shown in Figure 11. The disturbance rejection achieved is about 10 dB within a frequency range of 10-45 Hz, with some spillovers within a frequency of 45-60 Hz. Since the high frequency disturbances can be attenuated by passive damping, future work will evaluate the design of a passive layer to reduce the spillover effect.

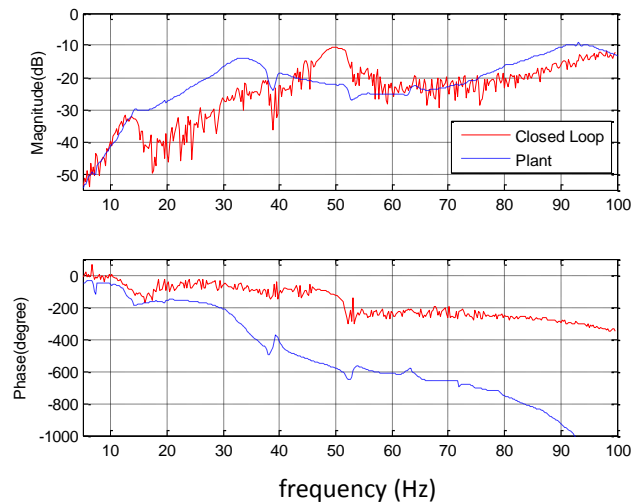


Figure 11. Comparison between the open loop and closed loop systems.

### Conclusion

A novel active-passive vibration isolation platform is proposed where the active layer in the vertical direction consists of a pendulum structure and a piezoelectric actuator with accelerometer embedded in the intermediate mass. On top of the active layer, a passive layer is designed to attenuate high frequency vibration. The isolation effect in the horizontal direction is purely passive. With the aid of pendulum structure, the first natural frequency can be reduced to about 1.5 Hz which can help attenuate high frequency disturbances. To understand the dynamic behavior of the system, the motion equations in the vertical and horizontal directions are derived using the Lagrangian approach. Device performance is validated by experiments and is used as the basis for controller design. To account for payload uncertainty, the robust  $H_\infty$  controller is adopted. The uncertainty weighting function is determined to cover payloads ranging from 40 Kg to 60 Kg. Experimental results show that the active control can reduce the vibration by an average of 10dB within the frequency range from 15-40Hz.

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Table 1. Parameters of vibration isolation module in horizontal and vertical directions.

|          |                |
|----------|----------------|
| $M_u$    | 50 Kg          |
| $M_d$    | 2.54           |
| $K_u$    | 2 N / m        |
| $K_{uh}$ | 0.57N/ m       |
| $K_z$    | 10 N / m       |
| $K_s$    | 1 N/ m         |
| $C_u$    | 1200 N.sec / m |
| $C_{uh}$ | 300 N.sec / m  |
| $C_z$    | 700 N.sec / m  |
| $C_s$    | 700 N.sec / m  |
| $C_{dh}$ | 90 N.sec / m   |
| $J_c$    | 0.1 N / V      |
| $c_h$    | 58 N sec / m   |
| $L_w$    | 0.0375 m       |

